



<http://elec3004.com>

Applications in Industry – (Extended) Kalman Filter

ELEC 3004: Systems: Signals & Controls
Dr. Surya Singh

Lecture 22
(with material from B. Lathi and Jur van den Berg)

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<http://robotics.itee.uq.edu.au/~elec3004/>

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Lecture Schedule:

Week	Date	Lecture Title
1	29-Feb	Introduction
	3-Mar	Systems Overview
2	7-Mar	Systems as Maps & Signals as Vectors
	10-Mar	Data Acquisition & Sampling
3	14-Mar	Sampling Theory
	17-Mar	Antialiasing Filters
4	21-Mar	Discrete System Analysis
	24-Mar	Convolution Review
	28-Mar	Holiday
	31-Mar	
5	4-Apr	Frequency Response & Filter Analysis
	7-Apr	Filters
6	11-Apr	Digital Filters
	14-Apr	Digital Filters
7	18-Apr	Digital Windows
	21-Apr	FFT
8	25-Apr	Holiday
	28-Apr	Introduction to Feedback Control
9	3-May	Holiday
	5-May	Feedback Control & Regulation
10	9-May	Servoregulation/PID
	12-May	Introduction to (Digital) Control
11	16-May	Digital Control Design & State-Space
	19-May	Observability, Controllability & Stability of Digital Systems
12	23-May	Digital Control Systems: Shaping the Dynamic Response & Estimation
	26-May	Applications in Industry
13	30-May	System Identification & Information Theory
	2-Jun	Summary and Course Review



Announcements

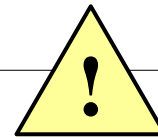
PS 3 Peer Review Competition

- The PS 3 Review with the highest Likert Score
- Deadline for reviews: June 3 (11:59 pm)
- Good reviews discussed June 2nd Last Lecture
- Reward: 3004¢

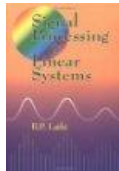


ELEC 3004 “Review Lab” (“Lab 5”):

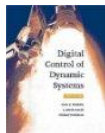
- Redo any aspects of any of the labs
- Review course
- Review 2015 Final exam (which we will do on June 2nd also)



Follow Along Reading:



B. P. Lathi
Signal processing and linear systems
1998
[TK5102.9.L38 1998](#)



**G. Franklin,
J. Powell,
M. Workman**
Digital Control of Dynamic Systems
1990

[TJ216.F72 1990](#)
[Available as
UQ Ebook]

Today

→ State-space ←

- FPW
 - Chapter 6 - Design of Digital Control Systems Using State-Space Methods

- Lathi Ch. 2 (?)
 - § 2.7-6 Time Constant and Rate of Information Transmission
- Information Theory!

Next Time



</assessable>

WARNING: NOT ASSESSABLE

- Nothing beyond this point is on the exam.
(except for the exam review 😊)
- Do not pay attention.
- Do not attempt to learn.



Example 0:
tf2ss

TF 2 SS – Control Canonical Form)

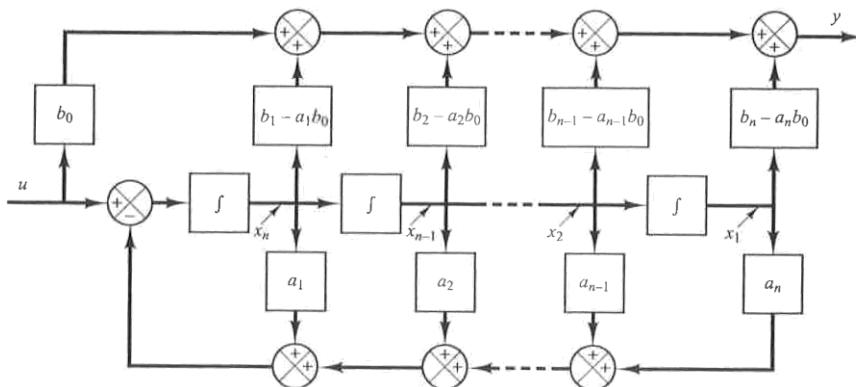
$$\frac{Y(s)}{U(s)} = \frac{b_0 s^n + b_1 s^{n-1} + \dots + b_{n-1} s + b_n}{s^n + a_1 s^{n-1} + \dots + a_{n-1} s + a_n}$$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \vdots \\ \dot{x}_{n-1} \\ \dot{x}_n \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & \dots & 0 \\ 0 & 0 & 1 & \dots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \dots & 1 \\ -a_n & -a_{n-1} & -a_{n-2} & \dots & -a_1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_{n-1} \\ x_n \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix} u$$

$$+ \quad y = [b_n - a_n b_0 \quad b_{n-1} - a_{n-1} b_0 \quad \dots \quad b_1 - a_1 b_0] \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} + b_0 u$$



Control Canonical Form as a Block Diagram



Modal Form

- CCF is not the only way to tf2ss
- Partial-fraction expansion of the system
 - System poles appear as diagonals of Am
- Two issues:
 - The elements of matrix maybe complex if the poles are complex
 - It is non-diagonal with repeated poles



Modal Form

$$\frac{Y(s)}{U(s)} = \frac{b_0 s^n + b_1 s^{n-1} + \dots + b_{n-1} s + b_n}{(s + p_1)(s + p_2) \dots (s + p_n)}$$

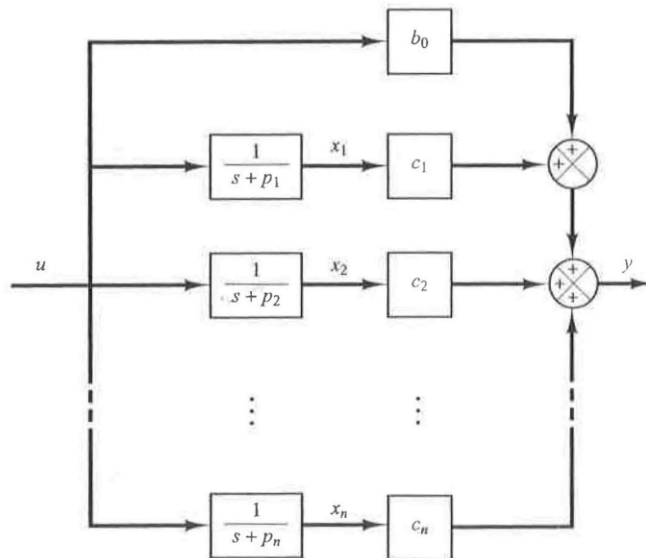
$$= b_0 + \frac{c_1}{s + p_1} + \frac{c_2}{s + p_2} + \dots + \frac{c_n}{s + p_n}$$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \vdots \\ \dot{x}_n \end{bmatrix} = \begin{bmatrix} -p_1 & & & 0 \\ & -p_2 & & \\ & & \ddots & \\ & & & -p_n \\ 0 & & & & \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} + \begin{bmatrix} 1 \\ 1 \\ \vdots \\ 1 \end{bmatrix} u$$

$$+ \quad y = [c_1 \quad c_2 \quad \dots \quad c_n] \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} + b_0 u$$



Modal Form Block Diagram



Matlab's tf2ss

- Given: $\frac{Y(s)}{U(s)} = \frac{25.04s+5.008}{s^3+5.03247s^2+25.1026s+5.008}$
Get a state space representation of this system

- Matlab:

```
num = [25.04 5.008];
den = [1 5.03247 25.1026 5.008];
[A,B,C,D] = tf2ss(num/den);
```

- Answer:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} -5.0325 & -25.1026 & -5.008 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} u$$

$$y = [0 \quad 25.04 \quad 5.008] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + [0]u$$

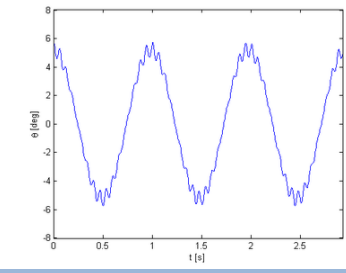
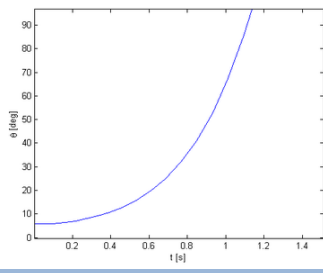


Example 1: Inverted Pendulum

Digital Control



Wikipedia,
Cart and pole



$$L = \frac{1}{2}M\dot{x}^2 + \frac{1}{2}m\dot{y}_2^2 - mgl \cos \theta$$

where v_1 is the velocity of the cart and v_2 is the velocity of the point mass m . v_1 and v_2 can be expressed in terms of x and θ by writing the velocity as the first derivative of the position:

$$v_1^2 = \dot{x}^2$$

$$v_2^2 = \left(\frac{d}{dt}(x - l \sin \theta)\right)^2 + \left(\frac{d}{dt}(l \cos \theta)\right)^2$$

Simplifying the expression for v_2 leads to:

$$v_2^2 = \dot{x}^2 - 2l\dot{\theta} \cos \theta + l^2\dot{\theta}^2$$

The Lagrangian is now given by:

$$L = \frac{1}{2}(M+m)\dot{x}^2 - m l \dot{\theta} \cos \theta + \frac{1}{2}m l^2 \dot{\theta}^2 - mgl \cos \theta$$

and the equations of motion are:

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{x}} - \frac{\partial L}{\partial x} = F$$

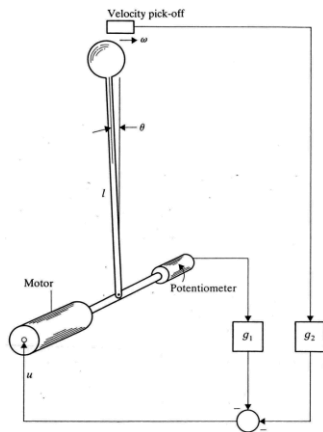
$$\frac{d}{dt} \frac{\partial L}{\partial \dot{\theta}} - \frac{\partial L}{\partial \theta} = 0$$

substituting L in these equations and simplifying leads to the equations that describe the motion

$$(M+m)\ddot{x} - m l \ddot{\theta} \cos \theta + m l \dot{\theta}^2 \sin \theta = F$$

$$l \ddot{\theta} - g \sin \theta = \ddot{x} \cos \theta$$

Inverted Pendulum



$$L = \frac{1}{2}Mv_1^2 + \frac{1}{2}mv_2^2 - mgl \cos \theta$$

where v_1 is the velocity of the cart and v_2 is the velocity of the point mass m . v_1 and v_2 can be expressed in terms of x and θ by writing the velocity as the first derivative of the position;

$$v_1^2 = \dot{x}^2$$

$$v_2^2 = \left(\frac{d}{dt}(x - \ell \sin \theta) \right)^2 + \left(\frac{d}{dt}(\ell \cos \theta) \right)^2$$

Simplifying the expression for v_2 leads to:

$$v_2^2 = \dot{x}^2 - 2\ell\dot{x}\dot{\theta} \cos \theta + \ell^2\dot{\theta}^2$$

The Lagrangian is now given by:

$$L = \frac{1}{2}(M + m)\dot{x}^2 - m\ell\dot{x}\dot{\theta} \cos \theta + \frac{1}{2}m\ell^2\dot{\theta}^2 - mgl \cos \theta$$

and the equations of motion are:

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{x}} - \frac{\partial L}{\partial x} = F$$

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{\theta}} - \frac{\partial L}{\partial \theta} = 0$$

substituting L in these equations and simplifying leads to the equations that describe the motion of

$$(M + m)\ddot{x} - m\ell\ddot{\theta} \cos \theta + m\ell\dot{\theta}^2 \sin \theta = F$$

$$\ell\ddot{\theta} - g \sin \theta = \ddot{x} \cos \theta$$

Inverted Pendulum – Equations of Motion

- The equations of motion of an inverted pendulum (under a small angle approximation) may be linearized as:

$$\dot{\theta} = \omega$$

$$\dot{\omega} = \ddot{\theta} = Q^2\theta + Pu$$

Where:

$$Q^2 = \left(\frac{M + m}{Ml} \right) g$$

$$P = \frac{1}{Ml}$$

If we further assume unity Ml ($Ml \approx 1$), then $P \approx 1$

Inverted Pendulum –State Space

- We then select a state-vector as:

$$\mathbf{x} = \begin{bmatrix} \theta \\ \omega \end{bmatrix}, \text{ hence } \dot{\mathbf{x}} = \begin{bmatrix} \dot{\theta} \\ \dot{\omega} \end{bmatrix} = \begin{bmatrix} \omega \\ \dot{\omega} \end{bmatrix}$$

- Hence giving a state-space model as:

$$A = \begin{bmatrix} 0 & 1 \\ Q^2 & 0 \end{bmatrix}, B = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

- The resolvent of which is:

$$\Phi(s) = (sI - A)^{-1} = \begin{bmatrix} s & -1 \\ -Q^2 & s \end{bmatrix}^{-1} = \frac{1}{s^2 - Q^2} \begin{bmatrix} s & 1 \\ Q^2 & s \end{bmatrix}$$

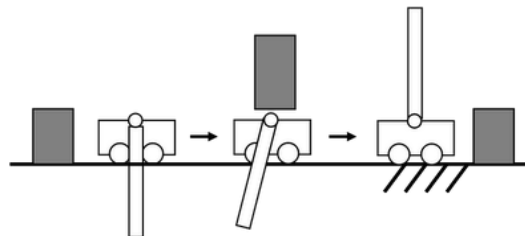
- And a state-transition matrix as:

$$\Phi(t) = \begin{bmatrix} \cosh Qt & \frac{\sinh Qt}{Q} \\ Q \sinh Qt & \cosh Qt \end{bmatrix}$$



Cart & Pole in State-Space With Obstacles?

Swing-up is a little more than stabilization...



See also: METR4202 – Tutorial 11:

<http://robotics.itee.uq.edu.au/~metr4202/tp1/t11-Week11-pendulum.pdf>



Cart & Pole in State-Space

Swing-up is a little more than stabilization...

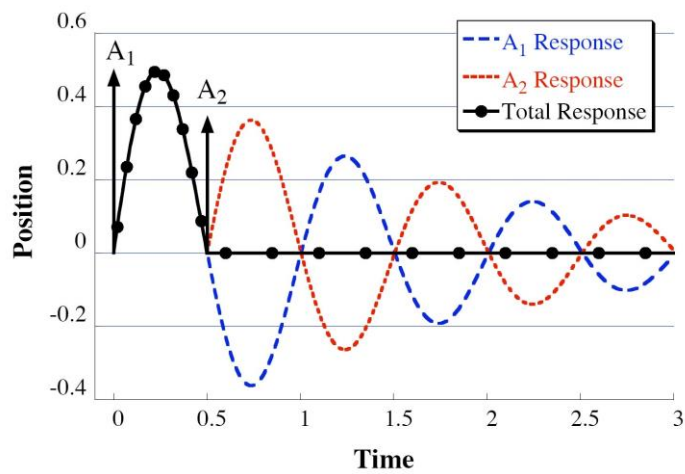


Example 2: Command Shaping

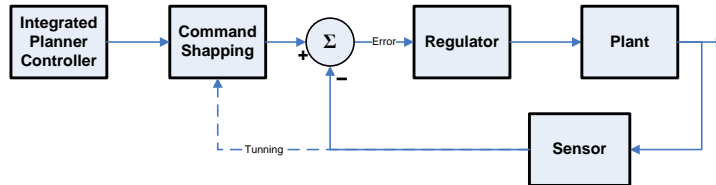
Experiments: Scanning Over Obstacle



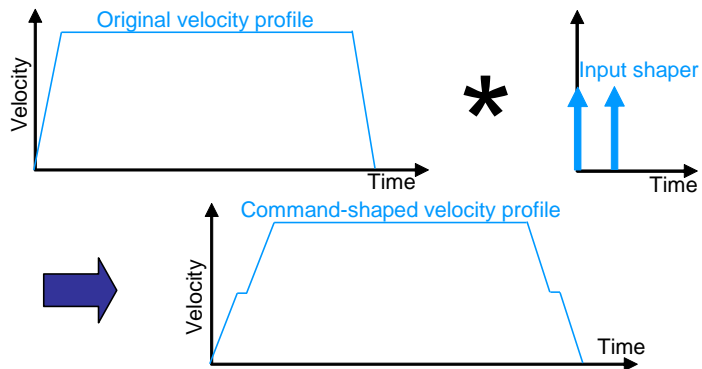
Command Shaping



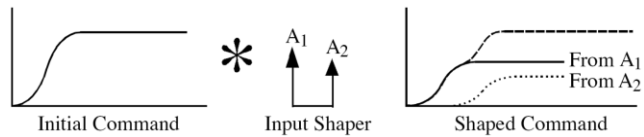
Robust Control: Command Shaping for Vibration Reduction



Command Shaping



Command Shaping



- Zero Vibration (ZV)

$$\begin{bmatrix} A_i \\ t_i \end{bmatrix} = \begin{bmatrix} 1 & K \\ 1+K & 1+K \\ 0 & \frac{T_d}{2} \end{bmatrix} \quad K = e^{\left(\frac{-\zeta\pi}{\sqrt{1-\zeta^2}} \right)}$$

- Zero Vibration and Derivative (ZVD)

$$\begin{bmatrix} A_i \\ t_i \end{bmatrix} = \begin{bmatrix} 1 & 2K & K^2 \\ (1+K)^2 & (1+K)^2 & (1+K)^2 \\ 0 & \frac{T_d}{2} & T_d \end{bmatrix}$$



Experiments: Command Shaping



Estimation: Yet another way to beat the noise

Along multiple dimensions



State Space

- We collect our set of uncertain variables into a vector ...
 $\mathbf{x} = [x_1, x_2, \dots, x_N]^T$
- The set of values that \mathbf{x} might take on is termed the *state space*
- There is a *single* true value for \mathbf{x} ,
but it is unknown



State Space Dynamics

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}$$

$$\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u}$$

$$H(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}$$



Measured versus True

- Measurement errors are inevitable
- So, add Noise to State...
 - State Dynamics becomes:

$$\dot{x} = Ax + Bu + w$$

$$y = Cx + Du + v$$

- Can represent this as a Normal Distribution

$$\mathcal{N}(x; \mu, \sigma) = \frac{1}{(\sqrt{2\pi}) \sigma} \exp\left(-\frac{(x - \mu)^2}{2\sigma^2}\right)$$



Recovering The Truth

- Numerous methods
- Termed “Estimation” because we are trying to estimate the truth from the signal
- A strategy discovered by Gauss
- Least Squares in Matrix Representation

$$\begin{bmatrix} p_0 \\ p_1 \end{bmatrix} = \begin{bmatrix} n & \sum_1^n t_i \\ \sum_1^n t_i & \sum_1^n t_i^2 \end{bmatrix}^{-1} \begin{bmatrix} \sum_1^n z_i \\ \sum_1^n t_i z_i \end{bmatrix}$$



Recovering the Truth: Terminology

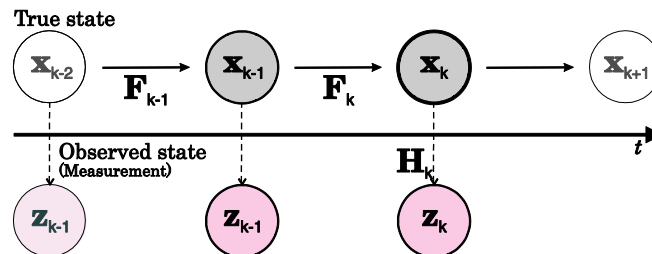
$$\dot{\mathbf{x}} = \mathbf{F}\mathbf{x} + \mathbf{G}\mathbf{u} + \mathbf{w}$$

$$\mathbf{z} = \mathbf{H}\mathbf{x} + \mathbf{v}$$

- \mathbf{x} : the state vector
- $\mathbf{x}_{A|B}$: the state of \mathbf{x} at time A based on data taken up to time B
- $\hat{\mathbf{x}}$: estimate of the true state vector
- \mathbf{F} : system dynamics matrix in continuous time (equivalent to \mathbf{A} in Eq. 1)
- \mathbf{G} : system control matrix relating deterministic input, \mathbf{u} , to the state (equivalent to \mathbf{B} in Eq. 1)
- \mathbf{H} : measurement matrix in continuous time (equivalent to \mathbf{C} in Eq. 2)
- \mathbf{F}_i : system model in **discrete** time at $t = t_i$
- \mathbf{H}_i : measurement model in **discrete** time at $t = t_i$
- \mathbf{P}_i : estimate covariance in **discrete** time at $t = t_i$
- \mathbf{w} : process uncertainty (noise) vector (of type $\mathcal{N}(0, s)$)
- \mathbf{Q} : process noise matrix, $\mathbf{Q} = E[\mathbf{w}\mathbf{w}^T]$
- \mathbf{Q}_i : \mathbf{Q} in discrete time at $t = t_i$
- \mathbf{v} : measurement noise vectors (of type $\mathcal{N}(0, \sigma)$)
- \mathbf{R}_i : the measurement variance matrix, $\mathbf{R} = E[\mathbf{v}\mathbf{v}^T]$, in discrete time at $t = t_i$



General Problem...

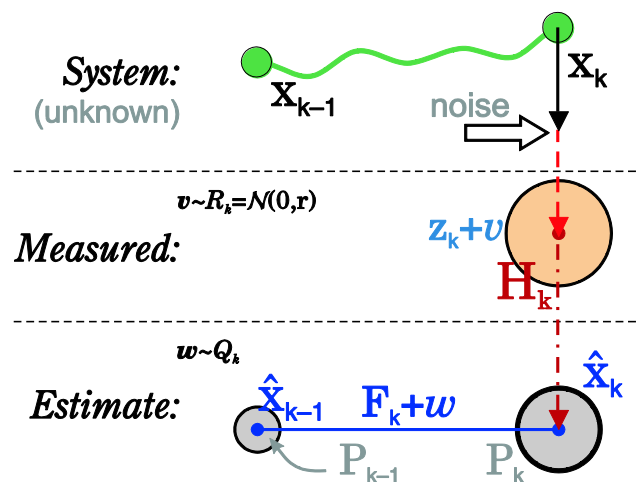


Duals and Dual Terminology

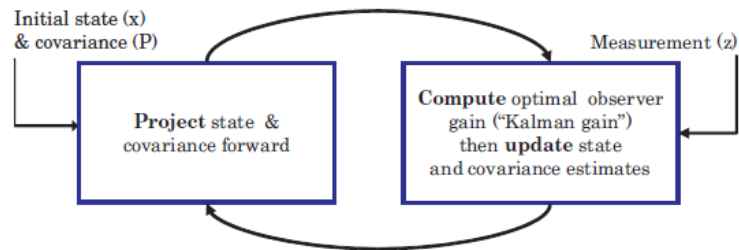
	Estimation		Control
Model:	$\dot{\mathbf{x}} = \mathbf{F}\mathbf{x}$ (discrete: $\mathbf{x} = \mathbf{F}_k\mathbf{x}$)	\leftrightarrow	$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x}$, $\mathbf{A} = \mathbf{F}^1$
Regulates:	\mathbf{P} (covariance)	\leftrightarrow	\mathbf{M} (performance matrix)
Minimized function:	Q (or GQG^1)	\leftrightarrow	V
Optimal Gain:	K	\leftrightarrow	G
Completeness law:	Observability	\leftrightarrow	Controllability



Estimation Process in Pictures



Kalman Filter Process



KF Process in Equations

$$\begin{aligned}
 \text{Prediction: } \hat{\mathbf{x}}_{k|k-1} &= \mathbf{F}_{k-1} \hat{\mathbf{x}}_{k-1|k-1}, && \text{(state prediction)} \\
 \mathbf{P}_{k|k-1} &= \mathbf{Q}_{k-1} + \mathbf{F}_{k-1} \mathbf{P}_{k-1|k-1} \mathbf{F}_{k-1}^T, && \text{(covariance prediction)} \\
 \text{Kalman Gain: } \mathbf{K}_k &= \mathbf{P}_{k|k-1} \mathbf{H}^T [\mathbf{H} \mathbf{P}_{k|k-1} \mathbf{H}^T + \mathbf{R}_k]^{-1}, \\
 \text{Update: } \mathbf{P}_{k|k} &= [\mathbf{I} - \mathbf{K}_k \mathbf{H}] \mathbf{P}_{k|k-1}, && \text{(covariance update)} \\
 \hat{\mathbf{x}}_{k|k} &= \hat{\mathbf{x}}_{k|k-1} + \mathbf{K}_k (\mathbf{z}_k - \mathbf{H} \hat{\mathbf{x}}_{k|k-1}) && \text{(state update)}
 \end{aligned}$$



KF Considerations

$$\begin{aligned} \hat{\mathbf{x}}_{k|k-1} &= \underbrace{\mathbf{F}_{k-1}}_{n \times n} \hat{\mathbf{x}}_{k-1|k-1} + \underbrace{\mathbf{G}_{k-1}}_{n \times j} \underbrace{\mathbf{u}_{k-1}}_{j \times 1} \\ \mathbf{P}_{k|k-1} &= \underbrace{\mathbf{Q}_{k-1}}_{n \times n} + \mathbf{F}_{k-1} \mathbf{P}_{k-1|k-1} \mathbf{F}_{k-1}^T \\ \mathbf{K}_k &= \underbrace{\mathbf{P}_{k|k-1}}_{n \times m} \underbrace{\mathbf{H}^T}_{n \times m} \underbrace{[\mathbf{H} \mathbf{P}_{k|k-1} \mathbf{H}^T + \mathbf{R}_k]^{-1}}_{m \times m} \\ \mathbf{P}_{k|k} &= [\mathbf{I} - \mathbf{K}_k \mathbf{H}] \mathbf{P}_{k|k-1} \\ \hat{\mathbf{x}}_{k|k} &= \hat{\mathbf{x}}_{k|k-1} + \mathbf{K}_k \left(\underbrace{\mathbf{z}_k}_{m \times 1} - \underbrace{\mathbf{H}}_{m \times n} \hat{\mathbf{x}}_{k|k-1} - \mathbf{H} \mathbf{G}_k \mathbf{u}_{k-1} \right) \end{aligned}$$



Ex: Kinematic KF: Tracking

- Consider a System with Constant Acceleration

$$\begin{aligned} \ddot{y} &= -g \\ \dot{y} &= gt + p_1 \\ y &= p_0 + p_1 t + \frac{gt^2}{2} \end{aligned}$$

$$\begin{bmatrix} \dot{y} \\ \ddot{y} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} y \\ \dot{y} \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} 0 \\ g \end{bmatrix}$$

$$\mathbf{F} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & \ddots & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 \end{bmatrix} \quad \mathbf{F}_k = \begin{bmatrix} 0 & t_s & \frac{t_s^2}{2} \\ 0 & 0 & t_s \\ 0 & 0 & 0 \end{bmatrix}$$

$$\hat{\mathbf{x}}_k = \mathbf{F}_{k-1} \hat{\mathbf{x}}_{k-1} + \mathbf{K}_k (\mathbf{z}_k - \mathbf{H} \mathbf{F}_{k-1} \hat{\mathbf{x}}_{k-1})$$



In Summary

- KF:
 - The true state (x) is separate from the measured (z)
 - Lets you **combine** prior controls knowledge with measurements to filter signals and find the truth
 - It **regulates** the covariance (P)
 - As P is the scatter between z and x
 - So, if $P \rightarrow 0$, then $z \rightarrow x$ (measurements \rightarrow truth)
- EKF:
 - Takes a Taylor series approximation to get a local “F” (and “G” and “H”)



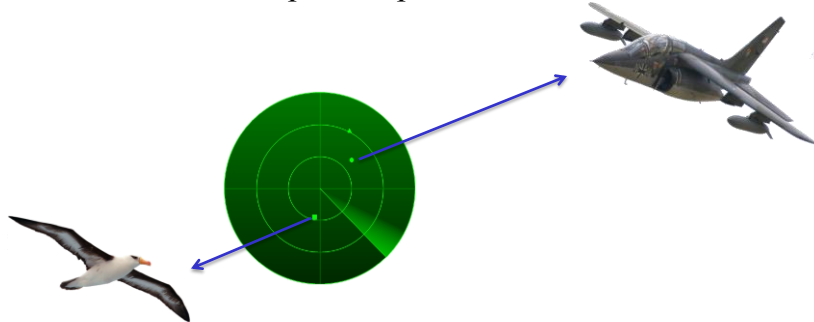
Estimation: “Bayesian Perspective”



$$P(A|B) = \frac{P(B|A)P(A)}{P(B)}$$

Kalman Filtering

- (Optimal) estimation of the (hidden) state of a linear dynamic process of which we obtain noisy (partial) measurements
- Example: radar tracking of an airplane.
What is the state of an airplane given noisy radar measurements of the airplane's position?



Model

- Discrete time steps, continuous state-space
- (Hidden) state: \mathbf{x}_t , measurement: \mathbf{y}_t
- Airplane example:
- Position, speed and acceleration

$$\mathbf{x}_t = \begin{pmatrix} x_t \\ \dot{x}_t \\ \ddot{x}_t \end{pmatrix}, \quad \mathbf{y}_t = (\tilde{x}_t)$$



Dynamics and Observation model

- Linear dynamics **model** describes relation between the state and the next state, and the observation:
- Airplane example (if process has time-step δ):

$$\mathbf{x}_{t+1} = A\mathbf{x}_t + \mathbf{w}_t, \quad \mathbf{w}_t \sim W_t = N(\mathbf{0}, Q)$$

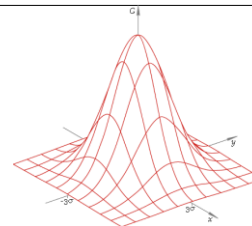
$$\mathbf{y}_t = C\mathbf{x}_t + \mathbf{v}_t, \quad \mathbf{v}_t \sim V_t = N(\mathbf{0}, R)$$

$$A = \begin{pmatrix} 1 & \delta & \frac{1}{2}\delta^2 \\ 0 & 1 & \delta \\ 0 & 0 & 1 \end{pmatrix}, \quad C = (1 \ 0 \ 0)$$



Normal distributions

- Let X_0 be a normal distribution of the initial state \mathbf{x}_0
- Then, every X_t is a normal distribution of hidden state \mathbf{x}_t . Recursive definition:



- And every Y_t is a normal distribution of observation \mathbf{y}_t .

Definition:

$$X_{t+1} = AX_t + W_t$$

- **Goal of filtering:** compute conditional distribution

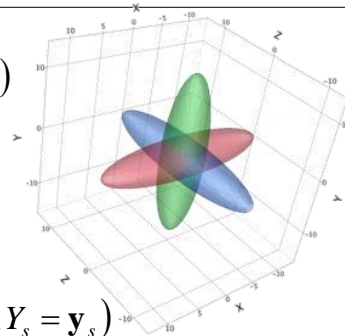
$$Y_t = CX_t + V_t$$

$$(X_t | Y_0 = \mathbf{y}_0, \dots, Y_t = \mathbf{y}_t)$$



Normal distribution

- Because X_t 's and Y_t 's are normal distributions, $(X_t | Y_0 = \mathbf{y}_0, \dots, Y_t = \mathbf{y}_t)$ is also a normal distribution
- Normal distribution is fully specified by mean and covariance
- We denote:



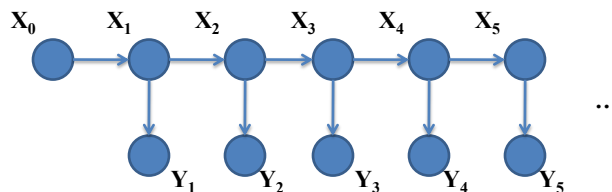
$$\begin{aligned}
 X_{t|s} &= (X_t | Y_0 = \mathbf{y}_0, \dots, Y_s = \mathbf{y}_s) \\
 &= N(\mathbf{E}(X_t | Y_0 = \mathbf{y}_0, \dots, Y_s = \mathbf{y}_s), \text{Var}(X_t | Y_0 = \mathbf{y}_0, \dots, Y_s = \mathbf{y}_s)) \\
 &= N(\hat{\mathbf{x}}_{t|s}, \mathbf{P}_{t|s})
 \end{aligned}$$

Problem reduces to computing $\hat{\mathbf{x}}_{t|t}$ and $\mathbf{P}_{t|t}$



Recursive update of state

- Kalman filtering algorithm: repeat...
 - Time update:
 - from $X_{t|t}$, compute **a priori** distribution $X_{t+1|t}$
 - Measurement update:
 - from $X_{t+1|t}$ (and given \mathbf{y}_{t+1}), compute **a posteriori** distribution $X_{t+1|t+1}$



Time update

- From $X_{t|t}$, compute **a priori** distribution $X_{t+1|t}$:

$$\begin{aligned}
 X_{t+1|t} &= AX_{t|t} + W_t \\
 &= N(\mathbf{E}(AX_{t|t} + W_t), \text{Var}(AX_{t|t} + W_t)) \\
 &= N(A\mathbf{E}(X_{t|t}) + \mathbf{E}(W_t), A\text{Var}(X_{t|t})A^T + \text{Var}(W_t)) \\
 &= N(A\hat{\mathbf{x}}_{t|t}, AP_{t|t}A^T + Q)
 \end{aligned}$$

- So:

$$\begin{aligned}
 \hat{\mathbf{x}}_{t+1|t} &= A\hat{\mathbf{x}}_{t|t} \\
 P_{t+1|t} &= AP_{t|t}A^T + Q
 \end{aligned}$$



Measurement update

From $X_{t+1|t}$ (and given \mathbf{y}_{t+1}), compute $X_{t+1|t+1}$.

1. Compute **a priori** distribution of the observation

$Y_{t+1|t}$ from $X_{t+1|t}$:

$$\begin{aligned}
 Y_{t+1|t} &= CX_{t+1|t} + V_{t+1} \\
 &= N(\mathbf{E}(CX_{t+1|t} + V_{t+1}), \text{Var}(CX_{t+1|t} + V_{t+1})) \\
 &= N(C\mathbf{E}(X_{t+1|t}) + \mathbf{E}(V_{t+1}), C\text{Var}(X_{t+1|t})C^T + \text{Var}(V_{t+1})) \\
 &= N(C\hat{\mathbf{x}}_{t+1|t}, CP_{t+1|t}C^T + R)
 \end{aligned}$$



Measurement update (cont'd)

2. Look at joint distribution of $X_{t+1|t}$ and $Y_{t+1|t}$:

$$\begin{aligned} (X_{t+1|t}, Y_{t+1|t}) &= N\left(\begin{pmatrix} \mathbf{E}(X_{t+1|t}) \\ \mathbf{E}(Y_{t+1|t}) \end{pmatrix}, \begin{pmatrix} \text{Var}(X_{t+1|t}) & \text{Cov}(X_{t+1|t}, Y_{t+1|t}) \\ \text{Cov}(Y_{t+1|t}, X_{t+1|t}) & \text{Var}(Y_{t+1|t}) \end{pmatrix}\right) \\ &= N\left(\begin{pmatrix} \hat{\mathbf{x}}_{t+1|t} \\ C\hat{\mathbf{x}}_{t+1|t} \end{pmatrix}, \begin{pmatrix} P_{t+1|t} & P_{t+1|t}C^T \\ CP_{t+1|t} & CP_{t+1|t}C^T + R \end{pmatrix}\right) \end{aligned}$$

where

$$\begin{aligned} \text{Cov}(Y_{t+1}, X_{t+1|t}) &= \text{Cov}(CX_{t+1|t} + V_{t+1}, X_{t+1|t}) \\ &= C \text{Cov}(X_{t+1|t}, X_{t+1|t}) + \text{Cov}(V_{t+1}, X_{t+1|t}) \\ &= C \text{Var}(X_{t+1|t}) \\ &= CP_{t+1|t} \end{aligned}$$



Measurement update (cont'd)

• Recall that if

$$(Z_1, Z_2) = N\left(\begin{pmatrix} \mu_1 \\ \mu_2 \end{pmatrix}, \begin{pmatrix} \Sigma_{11} & \Sigma_{12} \\ \Sigma_{21} & \Sigma_{22} \end{pmatrix}\right)$$

then

$$(Z_1 | Z_2 = \mathbf{z}_2) = N(\mu_1 + \Sigma_{12}\Sigma_{22}^{-1}(\mathbf{z}_2 - \mu_2), \Sigma_{11} - \Sigma_{12}\Sigma_{22}^{-1}\Sigma_{21})$$

3. Compute $X_{t+1|t+1} = (X_{t+1|t} | Y_{t+1|t} = \mathbf{y}_{t+1})$:

$$\begin{aligned} X_{t+1|t+1} &= (X_{t+1|t} | Y_{t+1|t} = \mathbf{y}_{t+1}) \\ &= N\left(\hat{\mathbf{x}}_{t+1|t} + P_{t+1|t}C^T(CP_{t+1|t}C^T + R)^{-1}(\mathbf{y}_{t+1} - C\hat{\mathbf{x}}_{t+1|t}), \right. \\ &\quad \left. P_{t+1|t} - P_{t+1|t}C^T(CP_{t+1|t}C^T + R)^{-1}CP_{t+1|t}\right) \end{aligned}$$



Measurement update (cont'd):

This can also (often) be written in terms of the **Kalman gain** matrix:

$$\begin{aligned}K_{t+1} &= P_{t+1|t} C^T (C P_{t+1|t} C^T + R)^{-1} \\ \hat{\mathbf{x}}_{t+1|t+1} &= \hat{\mathbf{x}}_{t+1|t} + K_{t+1} (\mathbf{y}_{t+1} - C \hat{\mathbf{x}}_{t+1|t}) \\ P_{t+1|t+1} &= P_{t+1|t} - K_{t+1} C P_{t+1|t}\end{aligned}$$



Initialization

- Choose distribution of initial state by picking \mathbf{x}_0 and P_0
- Start with measurement update given measurement \mathbf{y}_0
- Choice for Q and R (identity)
 - small Q: dynamics “trusted” more
 - small R: measurements “trusted” more



(Bayesian) Kalman Filter Summary

I. Model:

$$\mathbf{x}_{t+1} = A\mathbf{x}_t + \mathbf{w}_t, \quad \mathbf{w}_t \sim W_t = N(\mathbf{0}, Q)$$
$$\mathbf{y}_t = C\mathbf{x}_t + \mathbf{v}_t, \quad \mathbf{v}_t \sim V_t = N(\mathbf{0}, R)$$

II. Algorithm: Repeat...

– Time update:

$$\hat{\mathbf{x}}_{t+1|t} = A\hat{\mathbf{x}}_{t|t}$$
$$P_{t+1|t} = AP_{t|t}A^T + Q$$

– Measurement update:

$$K_{t+1} = P_{t+1|t}C^T(CP_{t+1|t}C^T + R)^{-1}$$
$$\hat{\mathbf{x}}_{t+1|t+1} = \hat{\mathbf{x}}_{t+1|t} + K_{t+1}(\mathbf{y}_{t+1} - C\hat{\mathbf{x}}_{t+1|t})$$
$$P_{t+1|t+1} = P_{t+1|t} - K_{t+1}CP_{t+1|t}$$



(Bayesian) Kalman Filter Summary [II]

Take Aways:

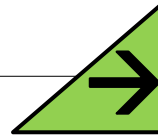
- Kalman filter can be used in real time
- Use $\hat{\mathbf{x}}_{t|t}$'s as optimal estimate of state at time t, and use $P_{t|t}$ as a measure of uncertainty.

Extensions:

- Dynamic process with known **control input**
- **Non-linear** dynamic process
- **Kalman smoothing**: compute optimal estimate of state \mathbf{x}_t given all data $\mathbf{y}_1, \dots, \mathbf{y}_T$, with $T > t$ (not real-time).
- Automatic parameter (Q and R) fitting using EM-algorithm



Next Time...




- **Information Theory & More!**
- **Review:**
 - Chapter 6 of FPW
 - Chapter 13 of Lathi
- **Deeper Pondering??**



Final Exam Review

- June 10, 2016
- From: 2-4
- In: 45-204 (???)
- [Some Review Notes \(from Course Textbooks\)](#)
→ <http://robotics.itee.uq.edu.au/~elec3004/tutes.html>

Semester One Final Examinations, 2016 ELEC3004 Signals, Systems & Control

 Venue: _____
Seat Number: _____
Student Number: _____
Family Name: _____
First Name: _____

This exam paper must not be removed from the venue

School of Information Technology and Electrical Engineering
EXAMINATION
Semester One Final Examinations, 2016
ELEC3004 Signals, Systems & Control
This paper is for St Lucia Campus students

Examination Duration: 180 minutes
Reading Time: 10 minutes

Exam Conditions:
This is a Closed Examination
This is a Closed Book Examination - specified materials permitted
During reading time - write only on the rough paper provided
This examination paper will be released to the Library

Materials Permitted in The Exam Venue:
(No electronic aids are permitted e.g. laptops, phones)
Any unmarked paper dictionary is permitted
An unmarked bilingual dictionary is permitted
Calculators - Any calculator permitted - unrestricted
One A4 sheet of handwritten or typed notes double sided is permitted

Materials To Be Supplied To Students:
1 x 14 Page Answer Booklet
1 x 1cm x 7cm Graph Paper
Rough Paper

Instructions To Students:
Additional exam materials (eg. answer booklet, rough paper) will be provided upon request.

Question	Mark
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Total: _____

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